



# WCFC Piper PA28 Warrior Quiz

Review before : 2023-01-05

Quiz ID : 2046

Instructor :	Date :
Pilot :	Score :

**Instructor :** Please note the final score (subtract 3.0 points from 100 for each wrong answer) on the checkout form and file the quiz in the Pilot Records folder.

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1 : The rated power of the engine as installed in a PA-28-161 is

- A : 100 BHP
- B : 125 BHP
- C : 150 BHP
- D : 160 BHP

2 : Best rate of climb speed for the PA-28-161 at gross weight at sea level is

- A : 52 KIAS
- B : 73 KIAS
- C : 79 KIAS
- D : 85 KIAS

3 : In the combination G5 and G500 autopilot installed in the WCFC PA28.161 Warriors, once engaged, the torque applied by ESP is at its maximum when bank angle ...

- A : exceeds a prudent bank angle for more than 5 seconds
- B : exceeds a roll rate of 45 degrees per second
- C : is accompanied by a loss or gain of more than 175 feet of altitude
- D : is 15 degrees more than the configured bank limit.
- E : the trim malfunctions and produces a runaway condition requiring immediate disabling of the electric trim

4 : If an engine fire should occur in flight, which are the recommended actions?

- A : Fuel selector OFF. Throttle closed. Mixture idle cut-off. Electric fuel pump OFF. Heater and defroster OFF. Cowl flaps open. If radio communication is not required, master switch OFF. Proceed with a power off landing.
- B : Fuel selector OFF. Throttle closed. Mixture idle cut-off. Electric fuel pump OFF. Heater and defroster OFF. If radio communication is not required, master switch OFF. Proceed with a power off landing.
- C : Fuel selector OFF. Mixture idle cut-off. Throttle open. Electric fuel pump OFF. Heater and defroster OFF. If radio communication is not required, master switch OFF. Proceed with a power off landing.
- D : Fuel selector OFF. Throttle closed. Mixture idle cut-off. Primer closed and latched. Crank the starter to induct the fire. Heater and defroster OFF. If radio communication is not required, master switch OFF. Proceed with a power off landing.
- E : Fuel selector OFF. Throttle closed. Mixture idle cut-off. Electric fuel pump OFF. Heater and defroster OFF. Cabin vents OPEN. If radio communication is not required, master switch OFF. Proceed with a power off landing.

5 : The correct type of fuel for the PA-28-161 (excepting any special STC) is

- A : Aviation 80, 100LL, or 100/130 fuel
- B : Automotive high test
- C : Aviation 100LL or 100/130 fuel (100LL preferred)
- D : Aviation 100LL (light blue) fuel only

6 : According to this representation of the G5 is Electronic Stability Protection enabled or disabled?



- A : Disengaged
- B : Enabled
- C : Disabled
- D : Standby

7 : When the ESP system that is an integral part of the G5 and the GFC 500 AFCS has been engaged for more than 10 seconds (cumulative; not necessarily consecutive seconds) of a 20-second interval, what happens?

- A : A warning alert "Pitch down" is heard.
- B : Flight controls are locked for five seconds to prevent further excursions in pitch and bank
- C : The autopilot is immediately disengaged, returning control to the pilot for safety.
- D : The autopilot engages in Level (LVL) mode
- E : The ESP system disengages to prevent over-driving the pitch and roll servos.

8 : The engine in a PA-28-161 is a

- A : Continental O-300
- B : Lycoming O-320
- C : Lycoming O-235
- D : Lycoming O-540

9 : The presentation labeled #14 indicates what information?



- A : airspeed
- B : selected vertical speed
- C : vertical speed bug
- D : true airspeed
- E : Ground speed

10 : According to the Warrior AFM/POH, engine fires during starting are usually caused by...

- A : cranking the starter excessively, thus flooding the engine
- B : priming with the auxiliary boost pump
- C : attempting to start the engine with the magnetos energized
- D : over-priming
- E : allowing fuel to vaporize on a hot day

11 : What is the longest VFR flight (hours and minutes) that the WCFC SOPs allow to be planned for a PA-28-161 with fuel to the tabs under VFR?

- For private pilots?
- For student pilots?
- NOTE
- Fueled to tabs, the PA28-161 Warrior II carries 34 gallons of usable fuel
- Use Figure 5-16 from page 5-22 for fuel consumption.
- Assume Best Power mixture and fuel flow at 65%

A : 3:86 private pilot, 1:93 student pilot

B : 2:51 private pilot, 1:08 student pilot

C : 2:86 private pilot, 1:19 student pilot

D : 3:00 private pilot, 1:19 student pilot

12 : What does STC SA00397NY, installed in some of the WCFC Warrior fleet, change about a PA28-161.

- A : This STC approves the Garmin G5 electronic instrument to serve as the primary attitude indicator and fully replace the original vacuum-powered instrument.
- B : The STC allows the installation of a smaller-diameter nose wheel to reduce the angle of attack on the takeoff roll and reduce the tendency to lift off prematurely in ground effect.
- C : The STC grants permission to operate that serial number airplane at a maximum gross weight of 2440 pounds rather than the original 2325 pounds.
- D : The STC allows the relocation of the battery from the firewall to an alternate location beneath the rear seat, which moves the CG aft (but within limits) to reduce drag and improve speed, fuel efficiency, and range.

13 : Under normal circumstances (leaned) the PA-28-161 burns approximately [Edit ... more](#) specific ... best power / economy etc, percentage power

- A : 6.5 gph at cruise power
- B : 8.5 gph at cruise power
- C : 9.5 gph at cruise power
- D : 10.5 gph at cruise power

14 : The type of oil normally in the engine should be ...

- A : SAE rated SE (severe environment) multi-viscosity
- B : Aviation grade ashless dispersant (AD) of appropriate viscosity
- C : High quality automotive type high detergent (HD) motor oil
- D : Aviation grade "straight mineral oil"
- E : Aviation grade multi-viscosity synthetic oil

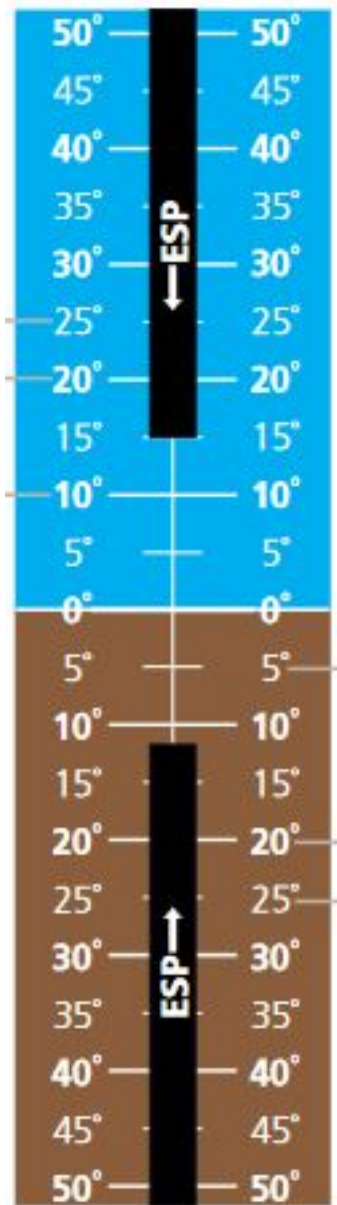
15 : Calculate the weight, CG, and total moment of N8080A using the data below. Choose the correct answer.

item	weight (pounds)	CG (arm)	Moment (/1000)
Airplane (80A)	1521.5	86.99	132.36
Front seat	220	80.5	17.71
Rear Seat	340	118.1	40.15
Fuel (pounds)	204	95.0	19.38
Baggage	100	142.8	14.28
Totals	-----	-----	-----

- A : Totals | 2385.5 | 90.23 | 223.88 |
- B : Totals | 2385.5 | 92.86 | 221.51 |
- C : Totals | 2585.5 | 93.86 | 242.67 |
- D : Totals | 2385.5 | 93.86 | 223.88 |

16 : What is the most current CG (Center of Gravity) in the WCFC record for N64TZ?

17 : If the configurable pitch limits, up and down, are both set to 20 degrees of pitch, the maximum torque applied by the ESP system will occur at what pitch levels?



- A : when the pilot attempts to override the pitch limits
- B : 20 degrees pitch up and 20 degrees pitch down
- C : 25 degrees pitch up and 25 degrees pitch down
- D : 15 degrees pitch up and 25 degrees down
- E : 15 degrees pitch up and 15 degrees down

18 : What would be the required fuel flow per hour and RPM setting to achieve 65% power at these parameters? (Use N8080A for performance calculations.)

- NOTE:

- Fuel to tabs (34 gallons usable)
- Wheel fairings installed
- Max Gross Weight 2325 pounds
- Best power mixture setting
- 65% power
- 8000 foot pressure altitude
- OAT (at altitude) 40F

A : 8.8 gph, 2530 RPM

B : 10 gph, 2530 RPM

C : 7.8 gph, 2580 RPM

D : 7.5 gph, 2600 RPM

19 :

- To achieve 65% power at 10,000 feet pressure altitude with an OAT of 20F requires what RPM setting? Assume best power mixture, leaned as indicated, and indicated fuel flow.
- Note that due to the inherent imprecision of these charts, choose the closest answer to your conclusion.

A : 2620 RPM

B : 2400 RPM

C : 2720 RPM

D : 2580 RPM

E : 2320 RPM



20 : This G5 is configured with a ...



- A : sky pointer
- B : ground pointer

21 : Using the Performance Charts of the AFM/POH for N8080A, serial number 18-8016051, calculate the true airspeed under the following conditions ... (Use the original maximum gross weight of 2325 pounds.)

Condition	Data
Cruise Pressure Altitude	8,000 feet
OAT at cruise altitude	15 Celsius
Cruise Power	65% best power
Wheel fairings	not installed
Gross weight	2325 pounds

- A : 105 knots TAS
- B : 109 knots TAS
- C : 122.5 knots TAS
- D : 125 knots TAS

22 : The following numbers refer to the various sources of information in the standard G5 PFD presentation as installed in the Club Warriors and integrated with the G500 Garmin autopilot. Consult the diagram and refer to the numbered items. Please fill in the blanks.  
 #15 is the \_\_\_\_\_. #17 is the \_\_\_\_\_. #14 is the \_\_\_\_\_. #19 is the \_\_\_\_\_ #20 is the \_\_\_\_\_.



23 : Recommended short-field flap setting, rotation speed, and initial-climb airspeeds for a best obstacle clearance takeoff for the PA-28-161 are, in order:

- A : 0 degrees, 50 KIAS, and 65 KIAS
- B : First notch of flaps, 63 KIAS, and 65 KIAS
- C : 25 degrees flaps, 52 KIAS, and 52 KIAS
- D : 40 degrees flaps, 52 KIAS, and 79 KIAS

24 : What would be a minimally inconvenient loading change that would suffice to make this aircraft legally flyable?

item	weight	CG (arm)	Moment (/1000)
Airplane (80A)	1521.5	86.99	132.36
Front seat	220	80.5	17.71
Rear Seat	340	118.1	40.15
Fuel (pounds)	204	95.0	19.38
Baggage	100	142.8	14.28
Totals	2385.5	93.86	223.88

- A : No change is necessary. Leave 50 lbs of baggage behind.
- B : Leave 50 lbs of baggage behind.
- C : Offload at least one passenger.
- D : Have the kid and one 170 lb passenger exchange seats.
- E : 2385.5 | 93.86 | 223.88

25 : If we suspect a total loss of alternator output in the PA28-161 electrical system, how can we determine if the output is a total loss of the alternator or merely a low demand on the system?

- A : Pull the alternator field circuit breaker to check for an increased reading on the ammeter
- B : Check the ammeter for a reading below zero that would indicate a failure and a battery discharge.
- C : Check the voltmeter for voltage of at least 14 volts, required for normal operation of the 12-volt system.
- D : Activate an electrically-powered system, such as the landing light, and if the ammeter does not respond, the alternator can be assumed to have failed.

26 : When flying an autopilot coupled LNAV approach with vertical descent angle and the MDA set as an altitude preselect, what will the autopilot do at MDA?

- A : The autopilot will level at the preselected MDA and continue to track the course guidance
- B : The autopilot will alert arrival at the MDA and request further command
- C : The autopilot will continue to descend on the vertical angle without leveling at the MDA
- D : The autopilot will disengage
- E : The autopilot will commence the missed approach

27 : The maximum gross takeoff weight for the CHFC PA-28-161 aircraft is

- A : 2000 pounds
- B : 2325 pounds
- C : 2350 pounds
- D : 2440 pounds

28 : At 2000 pounds total weight, a reasonable approximate maneuvering speed for the PA-28-161 is

- A : 76 KIAS
- B : 88 KIAS
- C : 102 KIAS
- D : 111 KIAS

29 : The active and armed modes, lateral and vertical, of the autopilot are displayed where?

- A : Active modes are displayed on the HSI function of the installed G5
- B : There is no separate display. The modes are recognized from the GFC 507 mode buttons, which turn red when engaged.
- C : Adjacent to the GFC 507 AFCS in the GFC 500 screen
- D : Autopilot (AP) status is displayed in the middle of the G5 Autopilot Status Box.
- E : The autopilot (AP) modes may be displayed externally on an Ipad linked via Bluetooth to the GFC 500 AFCS system.

30 : Given the following loadings, are the Normal category weight and balance limitations met for a typical CHFC Warrior? Use the N8080A AFM/POH Weight and Balance charts.

- For this calculation, please use these numbers for N8080A c. February 2001.
- Basic empty weight = 1521.5 pounds. CG = 86.99)
- Front seats: 150 lb pilot and 70 lb child passenger
- Rear seats: two 170 lb passengers
- Fuel: fueled to tabs
- Baggage: 100 lbs

This information in tabular format:

item	weight	CG (arm)	Moment (/1000)
Airplane (80A)	1521.5	86.99	132.36
Front seat	220	80.5	17.71
Rear Seat	340	118.1	40.15
Fuel (pounds)	204	95.0	19.38
Baggage	100	142.8	14.28
Totals	-----	-----	-----

- A : The weight is within STC limits, but the CG is too far aft.
- B : No. The weight is within STC limits, but the CG is too far forward.
- C : Yes. Both weight and CG are within STC limits.
- D : No. This airplane is over the STC maximum gross weight.
- E : Yes. But the CG is near the front limit.